



#### AVT-339 Research Workshop on Robotics and laser/plasma – paint interaction in paint removal

# Nd:YAG laser de-painting and its effects on military aircraft surfaces

#### Henning Baron Airbus Defence and Space GmbH Germany

29 – 30 April 2020







#### **Introduction: YAG:Nd laser de-painting process**

# Airbus has qualified two pulsed YAG:Nd laser de-painting processes:

- Handheld laser de-painting of electrical bonding points as per 80-M-35-0230
- Robot-guided laser de-painting of aircraft components as per 80-M-35-9140

Laser equipment parameters						
Manufacturer	Clean-Lasersysteme GmbH,					
	Herzogenaurach, Germany					
Main unit	CL500					
Scan frequency	100 Hz					
(applied)						
Scan width	≤ 70 mm					
Pulse frequency	27 kHz					
(applied)						
Brennweite	160 mm					
Focus diameter	700 μm (focal length of 150 mm)					
Laser class	4					
Feed rate	0.028 <sup>m</sup> / <sub>s</sub>					













#### Nd:YAG laser de-painting trial campaigns

- In order to gather hands-on experience and finally qualify the laser de-painting process, two de-painting trial campaigns were conducted by Airbus Manching.
- The first campaign focussed on the verification of process parameters and establishment of de-painting programs.
- The second campaign focussed on the potential of laser de-painting as a pre-treatment and adhesion promotion before re-painting.







#### **Carbon fibre composite specimens "Eurofighter"**



- The first group of specimens represents approx. 80 % of the EF exterior surface.
- Specimen built-up:

Polyurethane Top-Coat	<b>Alexit 472-22, matt</b> Thickness: 48 ± 5 μm		
Epoxy Primer, chromate-free	<b>Seevenax 113-24</b> Thickness: 22 ± 12 μm		
CFRP	Carbon Fibre prepreg Thickness: 2 mm		







#### Nd:YAG laser effects test programme



- $\geq$  50 µm of PUR top-coat is mostly removed after 3 de-painting overruns.
- $\geq$  25 µm of epoxy primer are still intact after the same energy intake.







#### **Carbon fibre composite specimens "Eurofighter"**



De-painted surfaces were examined by means of Scanning Electron Microscopy (SEM).



The examined CFRP specimen shows a very rough surface, some parts of which apparently being molten.







#### **Aluminium alloy specimens "Eurofighter"**



- The second group of specimens represents < 15 % of the EF exterior surface.
- Specimen built-up:

Polyurethane Top-Coat	<b>Alexit 472-22, matt</b> Thickness: 40 ± 10 μm		
Epoxy Primer, chromate-free	<b>Seevenax 113-24</b> Thickness: 25 ± 5 μm		
Epoxy Primer, chromate-loaded	<b>Seevenax 113-22</b> Thickness: 25 ± 5 μm		
Anodic Film	Thickness: 2 μm		
Aluminium alloy substrate	AA2024T3, unclad		









Amount of paint residues is clearly depending on the paint layer thickness: While there were only little residues on the 3 de-painting overrun specimen, on the 4-overrun-specimen up to 8  $\mu$ m of primer remain.







#### Aluminium alloy specimens "Eurofighter"

The "clean" specimen was examined in Energy-dispersive X-ray spectroscopy (EDX) for element analysis:





- > The shows traces of Barium (blue colour area), which originates from paint residues
- The coating material has not been entirely removed without residues.
- EDX shows an even distribution of oxygen compounds, which indicates the formation of an even oxide layer.



#### AVT-RWS-339





#### Nd:YAG laser effects test programme

#### Aluminium alloy specimens "Eurofighter"

The "clean" specimen was examined in Scanning Electron Microscopy (SEM).



- Both CAA substrates (pictures b and c) show the removal of the anodizing layer. Instead there is found an apparently compact oxide layer of 40 to 60 nm in thickness that shows molten areas.
- In addition the oxide layer of both specimens is covered with small agglomerated particles.



5 overruns

On the left SEM picture paint residues are clearly visible.





#### Aluminium alloy specimens "Tornado"



- The third group of specimens represents > 80 % of the TOR exterior surface.
- Specimen built-up:

Polyurethane Top-Coat	<b>Aerodur HF A 133D</b> Thickness: 40 ± 10 μm		
Epoxy Primer, chromate-free	<b>Seevenax 113-24</b> Thickness: 25 ± 5 μm		
Wash primer, chromate-loaded	<b>Celerol 913-21</b> Thickness: 10 ± 2 μm		
Aluminium alloy substrate	AA2024T3, unclad		







#### Nd:YAG laser effects test programme



- Laser treatment damages the adhesion of the EP primer on the wash primer.
- > PUR top-coat and EP primer are removed easily after the 3<sup>rd</sup> overrun.
- Selective de-painting of top-coat is not possible.
- Wash primer residues remain on the surface for a very long time.







#### Aluminium alloy specimens "Tornado"

The "clean" specimen was examined in Energy-dispersive X-ray spectroscopy (EDX) for element analysis:





- > The EDX picture shows an even oxide layer.
- In addition, the picture still shows remnants of carbon, indicating an incomplete removal of the coating material.







#### Aluminium alloy specimens "Tornado"

The "clean" specimen was examined in Scanning Electron Microscopy (SEM).





- The SEM pictures show a compact oxide layer with thicknesses between 50 to 100 nm.
- The surface shows a grainy structure. The paint apparently has been completely removed.
- Smooth areas as well as molten areas are visible.







#### **Examination of newly generated oxide layer**

- Both aluminium alloy specimen types showed the generation of a compact oxide layer on the Al surface.
- SEM analysis of these layers revealed thicknesses between 10 and 100 nm.
- In order to characterize the oxide layers as a potential pre-treatment for re-painting, Airbus performed further examinations, i.e.:
  - Adhesion of subsequent paint coatings,
  - Corrosion protection of bare Al surface,
  - Corrosion protection of painted Al surface.
- For this purpose clean unpainted Al specimens were treated with the same laser parameters.









#### **Examination of newly generated oxide layer**

#### Paint adhesion on laser-de-painted surfaces



Good paint adhesion, but water blisters after water immersion!







#### **Examination of newly generated oxide layer**

Corrosion pr	otection	21 hours of salt surau avansura	24 riburs or sait spray exposure acc. to ISO9227:	TSA Reference untreated	ce,	Laser-treated CAA	Laser-treated AA2024, bare	Laser de- painted	
Salt spray exposure 1000 hours acc. to ISO9227 Painted and scribed	TSA Reference	Laser-treated bare Al-sheets		Laser -treated anodized Al- sheets		Laser de-painted Al-sheets			
Aerowave 2001 + Aerowave 5001 Aerodur 2100 MgRP		Deint Crosser			Pai	int Creepage			
+ Aerodur 5000		Paint Creepag	e			Blisters			





#### Summary

- Laser de-painting of all examined paint materials proved to be feasible. However, different types of paint require different amounts of energy.
- For full removal of all paint residues from Al surfaces high enrgies (≤ 7 overruns) are required.
- Laser-treatment generates a very rough CFRP surface with partly molten areas.
- Laser-treatment destroys anodic films and generates a dense oxide layer of 10 to 100 nm in thickness on Al surfaces.
- Examinations of these oxide layers showed detrimental behavior with regard to adhesion of subsequent paint layers and corrosion protection.
- ➔ Therefore Airbus Defence and Space took the decision that the laser de-painting process may not be applied for the removal of an entire coating system from an aluminium surface. Instead, the top-coat on an Al component shall be selectively de-painted until the primer surface is reached.
- In case of full de-painting the oxide layer has to be removed, e.g. by abrading.







#### **Open questions**

- Does CO<sub>2</sub> lasers have a similar influence on Al surfaces?
- Which effects does laser de-painting have on surface corrosion?
  - Are these "oxide layers" also removed?
  - Is corrosion removal (e.g. by means of shot peening) still required?
- How does the generated oxide layer effect fatigue life of treated parts?







## Nd:YAG laser de-painting and its effects on military aircraft surfaces

Thank you

